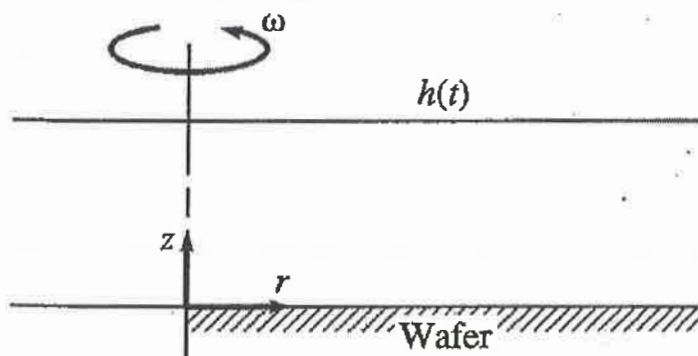


DRAFT v1 - SPIN COATING PROBLEM (ME 347 - FALL 2016)

Problem Statement:

In the microelectronics industry, very thin coatings of liquids are created on planar substrates by a process known as spin coating. The most common application of spin coating is to deposit photo-sensitive liquids, called photoresists, onto silicon wafers for pattern transfer. During the spin coating process, a liquid is first dispensed at the center of a substrate. The substrate is typically a flat circular surface. Following the dispense step, the substrate is rotated at a low speed (several hundred rpm) in order to spread the liquid uniformly across the surface. The film thickness at the end of this spread cycle is h_0 . Finally, the substrate is rotated at high speed (typically several thousand rpm) for a desired period of time to produce a thin film. During this spin cycle the film thickness is everywhere uniform; however, it is a function of time, i.e. $h(t)$ only. Your goal is to obtain an expression for the liquid film thickness h as a function of the spin speed ω , the spin time t , liquid viscosity μ , and liquid density ρ . Consider the problem to be one of axisymmetric laminar flow of a Newtonian fluid, where at any time during the spin cycle the liquid film is of uniform thickness everywhere on the substrate. Also consider that the substrate is uniformly covered with a thick layer of liquid prior to the spin, i.e., $h_0(t=0) \gg h(t)$. You may also assume that the viscosity of the liquid is such that viscous effects dominate over inertial effects. Note that although in practice the substrate has a finite diameter D , your solution will not depend on the extent of the substrate.

FIND

$$h(\omega, t, \mu, \rho)$$

ASSUME

AXISYMMETRIC

NEWTONIAN FLUID

LAMINAR FLOW

UNIFORM THICKNESS EVERYWHERE

* $h_0(t=0) \gg h(t)$ VISCUS EFFECTS DOMINATE INERTIAL $\Rightarrow Re$ is small

STANDARD ATMOSPHERE

INCOMPRESSIBLE

UNIFORM DENSITY

CONSTANT VISCOSITY

$$S = g \hat{e}_z$$

FIND

$h(t, w, \mu, \rho)$ or $h(t, w, \nu)$

ASSUME

- AXISYMMETRIC
- NEWTONIAN FLUID
- LAMINAR FLOW
- UNIFORM THICKNESS

$h_0(t=0) \gg h(t)$

VISCOSUS EFFECTS DOMINATE INERTIAL $\Rightarrow Re$ is small

STANDARD ATMOSPHERE

- INCOMPRESSIBLE
- UNIFORM DENSITY
- CONSTANT VISCOSITY
- $g = g \hat{e}_z$

IGNORE EVAPORATION

EQUATION OF MOTION: CYLINDRICAL NAVIER STOKES

- ONLY r-DIRECTION IS SIGNIFICANT

$$\rho \left(\frac{\partial v_r}{\partial t} + \underbrace{v_r \frac{\partial v_r}{\partial r}}_{\text{small continuity?}} + \frac{v_\theta}{r} \frac{\partial v_r}{\partial \theta} - \frac{v_\theta^2}{r} + \underbrace{v_z \frac{\partial v_r}{\partial z}}_{\text{possibly small}} \right)$$

change axisymmetric

$$= - \frac{\partial p}{\partial r} + \rho g \hat{e}_r + \mu \left[\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial v_r}{\partial r} \right) - \frac{v_r}{r^2} + \frac{1}{r^2} \frac{\partial^2 v_r}{\partial \theta^2} - \frac{2}{r^2} \frac{\partial v_\theta}{\partial \theta} + \frac{\partial^2 v_r}{\partial z^2} \right]$$

insignificant zero net in r direction negligible?

$-\rho \frac{v_\theta^2}{r} = \mu \frac{\partial^2 v_r}{\partial z^2}$

$v_\theta = \omega r$

$-\rho \frac{(\omega r)^2}{r} = \mu \frac{\partial^2 v_r}{\partial z^2}$

$-\rho \omega^2 r = \mu \frac{\partial^2 v_r}{\partial z^2}$



SECOND ORDER ORDINARY DIFFERENTIAL EQUATION WITH CONSTANT COEFFICIENTS

$$\frac{\partial^2 v_r}{\partial z^2} = -\frac{\rho}{\mu} \omega^2 r$$

SEPARATE INTEGRATE

$$\partial\left(\frac{\partial v_r}{\partial z}\right) = -\frac{\rho}{\mu} \omega^2 r \int \partial z$$

$$\frac{\partial v_r}{\partial z} = -\frac{\rho \omega^2 r}{\mu} [z + C_1]$$

APPLY BC ①

$$\frac{\partial v_r}{\partial z}(z=h) = -\frac{\rho \omega^2 r}{\mu} (h + C_1) = 0$$

$$\Rightarrow C_1 = -h$$

$$\frac{\partial v_r}{\partial z} = -\frac{\rho \omega^2 r}{\mu} (z - h)$$

SEPARATE INTEGRATE

$$\partial(v_r) = -\frac{\rho \omega^2 r}{\mu} \int (z - h) \partial z$$

$$v_r = -\frac{\rho \omega^2 r}{\mu} \left(\frac{z^2}{2} - hz + C_2 \right)$$

APPLY BC ②

$$v_r(z=0) = -\frac{\rho \omega^2 r}{\mu} (0 - 0 + C_2)$$

$$\Rightarrow C_2 = 0$$

$$\boxed{v_r(z) = -\frac{\rho \omega^2 r}{\mu} \left(\frac{z^2}{2} - hz \right)}$$

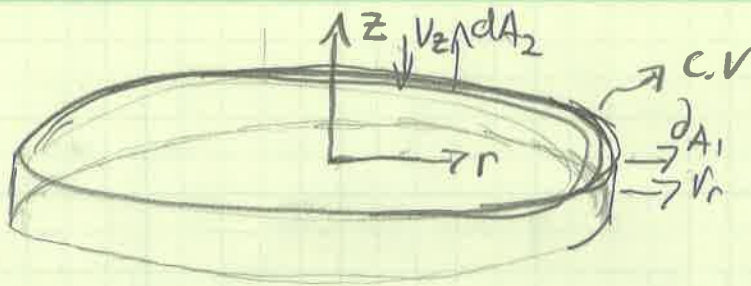
ASIDE: BOUNDARY CONDITIONS

$$\textcircled{1} \frac{\partial v_r}{\partial z}(z=h) = 0$$

free surface
no stress

$$\textcircled{2} v_r(z=0) = 0$$

surface
no slip



R.T.T OF MASS

$$\frac{\partial m}{\partial t} \Big|_{\text{sys}} = \frac{\partial}{\partial t} \int_{\text{CV}} \rho \, dV + \int_{A_1} \rho v_r \, dA_1 - \int_{A_2} \rho v_z \, dA_2$$

$$\rho \int v_r \, dA_1 = \rho \int v_z \, dA_2$$

$$2\pi R^2 v_z = 2\pi R \int_0^h \frac{\rho \omega^2 R}{\mu} \left(hz - \frac{z^2}{2} \right) dz$$

$$\pi R^2 v_z = 2\pi \frac{\rho \omega^2 R^2}{\mu} \int_0^h \left(hz - \frac{z^2}{2} \right) dz$$

$$= 2\pi \frac{\rho \omega^2 R^2}{\mu} \left[\frac{hz^2}{2} - \frac{z^3}{6} \right] \Big|_0^h$$

$$= 2\pi \frac{\rho \omega^2 R^2}{\mu} \left(\frac{h^3}{2} - \frac{h^3}{6} \right)$$

$$\pi R^2 v_z = \frac{2\pi \rho \omega^2 R^2 h^3}{3\mu}$$

$$v_z = \frac{2\rho \omega^2 h^3}{3\mu}$$

$$Vz = -\frac{dh}{dt}$$

SUBSTITUTE

$$= \frac{1}{h^3} dh = \frac{2}{3} \frac{\rho \omega^2}{\mu} dt$$

$$\int_{h_0}^h \frac{1}{h^3} dh = \int_0^t \frac{2}{3} \frac{\rho \omega^2}{\mu} dt$$

$$\frac{1}{h^2} \Big|_{h_0}^h = \frac{2}{3} \frac{\rho \omega^2}{\mu} \int dt$$

$$\frac{1}{2h^2} - \frac{1}{2h_0^2} = \frac{2}{3} \frac{\rho \omega^2}{\mu} t$$

* ASSUME $h_0(t=0) \gg h(t)$

$$h(t) = \sqrt{\frac{3\mu}{4\rho\omega^2 t}}$$

$$h(t) = \sqrt{\frac{3\mu}{4\rho\omega^2 t}}$$

7/24/16

SPIN COATING

ME-347 - RESEARCH

SAMUEL LEE

LAB PLAN

RESTRICTIONS

$\omega = 0 - 6000$ rpm

$t = 0 - 1000$ sec

Scale resolution 0.001 grams

Wafer diameter 2" - 6.5"

STEPS:

① CUT WAFERS

② MEASURE MASS OF WAFER

③ PLACE FLUID

④ SPIN

⑤ RE MEASURE FINAL MASS

⑥ CALCULATE PERCENT ERROR

EQUATIONS NAVIER STOKES
CONTINUITY EQUATION

$$\frac{\partial \rho}{\partial t} + \frac{1}{r} \frac{\partial (\rho r v_r)}{\partial r} + \frac{1}{r} \frac{\partial (\rho v_\theta)}{\partial \theta} + \frac{\partial (\rho v_z)}{\partial z} = 0 \quad (2)$$

EQUATION OF MOTION

r-direction

$$\rho \left(\frac{\partial v_r}{\partial t} + v_r \frac{\partial v_r}{\partial r} + \frac{v_\theta}{r} \frac{\partial v_r}{\partial \theta} - \frac{v_\theta^2}{r} + v_z \frac{\partial v_r}{\partial z} \right) = -\frac{\partial p}{\partial r} + \rho g_r + \mu \left[\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial v_r}{\partial r} \right) - \frac{v_\theta}{r^2} + \frac{1}{r^2} \frac{\partial^2 v_r}{\partial \theta^2} - \frac{2}{r^2} \frac{\partial v_\theta}{\partial \theta} + \frac{\partial^2 v_r}{\partial z^2} \right]$$

θ -direction

$$\rho \left(\frac{\partial v_\theta}{\partial t} + v_r \frac{\partial v_\theta}{\partial r} + \frac{v_\theta}{r} \frac{\partial v_\theta}{\partial \theta} + v_r v_\theta + v_z \frac{\partial v_\theta}{\partial z} \right) = -\frac{1}{r} \frac{\partial p}{\partial \theta} + \rho g_\theta + \mu \left[\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial v_\theta}{\partial r} \right) - \frac{v_\theta}{r^2} + \frac{1}{r^2} \frac{\partial^2 v_\theta}{\partial \theta^2} + \frac{2}{r^2} \frac{\partial v_r}{\partial \theta} + \frac{\partial^2 v_\theta}{\partial z^2} \right]$$

z-direction

$$\rho \left(\frac{\partial v_z}{\partial t} + v_r \frac{\partial v_z}{\partial r} + \frac{v_\theta}{r} \frac{\partial v_z}{\partial \theta} + v_z \frac{\partial v_z}{\partial z} \right) = -\frac{\partial p}{\partial z} + \rho g_z + \mu \left[\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial v_z}{\partial r} \right) + \frac{1}{r^2} \frac{\partial^2 v_z}{\partial \theta^2} + \frac{\partial^2 v_z}{\partial z^2} \right]$$

BOUNDARY CONDITIONS

- ① $v_r(z=0) = 0$ surface, no slip
- ② $\frac{\partial v}{\partial z}(z=h) = 0$ free surface, atmosphere

Neglect g_{rad} ? neglect pressure gradient?
so thin

THOUGHT CHECK

- STEP

- ① ASSUMPTIONS \Rightarrow NAVIER STOKES
- ② BOUNDARY CONDITIONS
- ③ CONTINUITY? MASS IS CONSERVED

$$\frac{\partial v}{\partial t} \quad \frac{\partial \rho}{\partial t} ?$$

WHAT CAN BE NEGLECTED?

1. GRAVITY? g_z REASON: THE ω (RPM) HAS A MUCH GREATER EFFECT

$$2. \frac{\partial p}{\partial z}$$

REASON: THIN, PRESSURE CHANGE IS MINIMAL ≈ 0

ASSUMPTIONS IN MORE DETAIL

① AXISYMMETRIC

WHY:

SIG:

② NEWTONIAN

WHY:

SIG:

③ LAMINAR

WHY:

SIG:

④ UNIFORM THICKNESS

WHY:

SIG:

⑤ $h_0(t=0) \gg h(t)$

WHY:

SIG:

⑥ VISCOUS EFFECTS DOMINATE INERTIAL

WHY

SIG:

ASSUMPTIONS CONTINUED

② STD. ATM

WHY:

SIG:

③ INCOMPRESSIBLE

WHY:

SIG:

④ UNIFORM DENSITY

WHY:

SIG:

⑤ CONSTANT VISCOSITY

WHY:

SIG:

⑥ $g = g \hat{e}_z$

WHY:

SIG:

$\frac{\partial p}{\partial t}$ does not change with time

$\frac{\partial v_\theta}{\partial \theta}$ constant

$\frac{\partial v_r}{\partial t}$ constant

$\frac{\partial v_r}{\partial \theta}$ v_r does not change with θ

$\frac{\partial v_r}{\partial z}$ does not change with z , negligible

$\frac{\partial p}{\partial r}$ does not change with r

$\partial g_r \quad g_z = g_z$

r-directions

$$\rho \left(v_r \frac{\partial v_r}{\partial r} - \frac{v_\theta^2}{r} \right) = \mu \left[\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial v_r}{\partial r} \right) - \frac{v_\theta}{r^2} + \frac{\partial^2 v_r}{\partial z^2} \right]$$

θ -direction

$$\rho v_r \frac{\partial v_\theta}{\partial r} = \mu \left[-\frac{v_\theta}{r^2} \right] \quad \frac{\partial v_r}{\partial r} = \frac{\mu}{r} \quad \frac{\partial v_\theta}{\partial r} \text{ not a}$$

z -direction

$$\rho v_z \frac{\partial v_z}{\partial z} = \mu \left[\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial v_z}{\partial r} \right) \right] \Rightarrow \frac{\partial v_z}{\partial z} = 0 \quad \textcircled{1}$$

Continuity

$$\frac{1}{r} \frac{\partial (r v_r)}{\partial r} + \frac{\partial v_z}{\partial z} = 0$$

$$\frac{1}{r} \frac{\partial (r v_r)}{\partial r} = 0 \quad \textcircled{2}$$

z direction

$$\rho \frac{\partial v_z}{\partial t} + \rho v_z \frac{\partial v_z}{\partial z} = 0$$

$$\frac{\partial v_z}{\partial t} = -v_z \frac{\partial v_z}{\partial z} \Rightarrow \frac{\partial v_z}{\partial z} = \frac{-v_z \partial v_z}{v_z \partial t}$$

$$v_r \frac{\partial v_r}{\partial r} + \frac{1}{r} \frac{\partial (r v_r)}{\partial r} + \frac{v_\theta}{r} \frac{d\theta}{dr} = 0$$

$$\frac{\partial v_r}{\partial r} + \frac{v_r}{r} \frac{\partial v_r}{\partial r} = 0$$

$$-\rho \frac{v_\theta^2}{r} = \mu \frac{\partial^2 v_r}{\partial z^2} \quad v_\theta = \omega r$$

$$-\rho \omega^2 r = \mu \frac{\partial^2 v_r}{\partial z^2}$$

separate and integrate

$$-\frac{1}{\mu} \rho \omega^2 r \int dz = \rho \left(\frac{\partial v_r}{\partial z} \right)$$

$$\frac{\partial v_r}{\partial z} = -\frac{1}{\mu} \rho \omega^2 r (z + c_1)$$

$$\text{BC (2)} \quad \frac{\partial v_r}{\partial z} (z=h) = 0$$

$$0 = -\frac{1}{\mu} \rho \omega^2 r (h + c_1) \Rightarrow$$

$$c_1 = -h$$

$$\frac{\partial v_r}{\partial z} = -\frac{1}{\mu} \rho \omega^2 r (z + h)$$

$$v_r = -\frac{1}{\mu} \rho \omega^2 r \left(\frac{z^2}{2} + hz + c_2 \right)$$

$$\text{BC (1)} \quad v_r(0) = 0$$

$$0 = \left(\frac{z^2}{2} + hz + c_2 \right) \quad c_2 = 0$$

$$v_r = -\frac{1}{\mu} \rho \omega^2 r \left(\frac{z^2}{2} + hz \right)$$

$$v_r = -\frac{\rho}{\mu} \omega^2 r \left(\frac{z^2}{2} + hz \right)$$

CONTINUITY

$$\frac{1}{r} \frac{\partial (r v_r)}{\partial r} + \frac{\partial v_z}{\partial z} = 0$$

$$\frac{\partial h}{\partial t} \quad ?$$

$$h = \omega r^2 \nu$$

$$\frac{\partial h}{\partial t} = \omega \frac{\partial (r^2 \nu)}{\partial t} = \omega r^2 \frac{\partial \nu}{\partial t} + 2\omega r \nu \frac{\partial r}{\partial t}$$

$$\frac{\partial h}{\partial t} = \omega r^2 \frac{\partial \nu}{\partial t} + 2\omega r \nu \frac{\partial r}{\partial t}$$

CONTINUITY

$$0 = \frac{1}{r} \frac{\partial(rv_r)}{\partial r} + \frac{\partial v_z}{\partial z} \approx \frac{1}{r} \frac{\partial(rv_r)}{\partial r}$$

$$\rho \left(rv_r \frac{\partial v_r}{\partial r} - \frac{v_r^2}{r} \right) = \mu \left[\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial v_r}{\partial r} \right) - \frac{v_r}{r^2} \right] + \frac{\partial^2 v_r}{\partial z^2}$$

$\star 0?$ $\star 0?$ \star

$$\boxed{\rho rv_r \frac{\partial v_r}{\partial r}} - \rho v_r^2 = \mu \frac{\partial^2 v_r}{\partial z^2}$$

$$Q = 2\pi r \int_0^h v_r \, dz = -\pi R^2 \frac{dh}{dt}$$

$$= 2\pi r \int_0^h \frac{\rho \omega^2 r}{\mu} \left(\frac{z^2}{2} + hz \right) dz$$

$$= -\frac{2\pi \rho \omega^2 r^2}{\mu} \left(\frac{z^3}{6} + h \frac{z^2}{2} \right) = -\pi R^2 \frac{dh}{dt}$$

$$= -\frac{2\pi \rho \omega^2 r^2}{\mu} \left(\frac{h^3}{6} + \frac{h^3}{2} \right) = -\frac{dh}{dt}$$

$$\left(\frac{h^3}{6} + \frac{3h^3}{6} \right)$$

$$= \frac{4}{3} \frac{\rho \omega^2 r^2}{\mu} h^3 = \frac{dh}{dt}$$

$$\frac{dh}{dt} = \frac{4}{3} \frac{\rho \omega^2 r^2}{\mu} h^3$$

$$\frac{1}{h^3} dh = \frac{4}{3} \frac{\rho \omega^2 r^2}{\mu} dt$$

$$h^{-3} \cdot \frac{h^{-2}}{-2} \Big|_{h_0}^{h_1}$$

$$h^{-2}$$

ENGINEERING DATA SHEET

Excel Spreadsheet

Sheet: Spin Coating Data

File : ME 347 - Spin Coating Project.xlsm



Sheet No. X - 1

Date

Prepared by: HCM

08/01/16

Checked by: XXX

XX/XX/XX

Approved by: XXX

XX/XX/XX

Data from spin coating experiments (H. Mayer & S. Lee)

| | | | | |
|------------|---------------------------|------------|-----|-------|
| Liquid: | Karo Syrup (DARK) | Sub. Dia.: | 100 | [mm] |
| Viscosity: | 3300 [mPa-s] | Time: | 50 | [sec] |
| Density: | 1370 [kg/m ³] | | | |

| Data No. [-] | Speed [rpm] | Bare Mass [gm] | Spun Mass [gm] | Liq. Mass [gm] | Thickness [um] | Unc.± [um] |
|-------------------|----------------|-------------------|-------------------|-------------------|-------------------|---------------|
| 1.1 | 3000 | 14.019 | 14.270 | 0.251 | 23.3 | 1.26 |
| 1.2 | 3000 | 13.602 | 13.850 | 0.248 | 23.0 | 1.24 |
| 1.3 | 3000 | 14.071 | 14.316 | 0.245 | 22.8 | 1.23 |
| 1.4 | 3000 | 13.977 | 14.224 | 0.247 | 23.0 | 1.24 |
| 1.5 | 3000 | 13.955 | 14.205 | 0.250 | 23.2 | 1.25 |
| 1.6 | 3000 | 13.883 | 14.133 | 0.250 | 23.2 | 1.25 |
| 1.7 | 3000 | 13.354 | 13.603 | 0.249 | 23.1 | 1.25 |
| 1.8 | 3000 | 13.853 | 14.103 | 0.250 | 23.2 | 1.25 |
| 2.1 | 6000 | 13.962 | 14.100 | 0.138 | 12.8 | 0.69 |
| 2.2 | 5000 | 14.041 | 14.178 | 0.137 | 12.7 | 0.69 |
| 2.3 | 4000 | 13.360 | 13.543 | 0.183 | 17.0 | 0.92 |
| 2.4 | 3000 | 13.984 | 14.217 | 0.233 | 21.7 | 1.17 |
| 2.5 | 2000 | 13.870 | 14.198 | 0.328 | 30.5 | 1.64 |
| 2.6 | 1000 | 13.888 | 14.544 | 0.656 | 61.0 | 3.28 |
| 3.1 | 6000 | 13.612 | 13.733 | 0.121 | 11.2 | 0.61 |

Notes:

1. Static charge from glove-substrate-paper towel contact can have a dramatic effect on the accuracy of measurements.
 Exampel, data point 2.1 is highly suspect and should be replaced with data point 3.1.

ENGINEERING DATA SHEET

Excel Spreadsheet
Sheet: Spin Coating Data
File : ME 347 - Spin Coating Project.xlsm



Sheet No. X - 2 Date
Prepared by: HCM 08/01/16
Checked by: XXX XX/XX/XX
Approved by: XXX XX/XX/XX

Data from spin coating experiments (H. Mayer & S. Lee)

Liquid: Canola Oil Sub. Dia.: 100 [mm]
Viscosity: 70 [mPa-s] Time: 50 [sec]
Density: 920 [kg/m^3]

| Data No. [-] | Speed [rpm] | Bare Mass [gm] | Spun Mass [gm] | Liq. Mass [gm] | Thickness [um] | Unc.± [um] |
|-------------------|----------------|-------------------|-------------------|-------------------|-------------------|---------------|
| 1.1 | 1000 | 13.609 | 13.689 | 0.080 | 11.1 | 0.60 |
| 1.2 | 5000 | 14.075 | 14.089 | 0.014 | 1.9 | 0.13 |
| 2.1 | 6000 | 13.797 | 13.810 | 0.013 | 1.8 | 0.12 |
| 2.2 | 5000 | 13.607 | 13.622 | 0.015 | 2.1 | 0.13 |
| 2.3 | 4000 | 14.076 | 14.099 | 0.023 | 3.2 | 0.18 |
| 2.4 | 3000 | 13.357 | 13.383 | 0.026 | 3.6 | 0.21 |
| 2.5 | 2000 | 13.968 | 13.998 | 0.030 | 4.2 | 0.23 |
| 2.6 | 1000 | 13.858 | 13.936 | 0.078 | 10.8 | 0.59 |
| 2.7 | 2000 | 13.886 | 13.924 | 0.038 | 5.3 | 0.29 |
| 2.8 | 1500 | 13.984 | 14.037 | 0.053 | 7.3 | 0.40 |

ENGINEERING DATA SHEET

Excel Spreadsheet
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Sheet No. X - 4 Date
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Approved by: XXX XX/XX/XX

HYPOTHETICAL thicknesses for future EXPERIMENTS

Liquid: Molassas Sub. Dia.: 100 [mm]
Viscosity: 10800 [mPa-s] Time: 50 [sec]
Density: 1400 [kg/m^3]

| Data No. [-] | Speed [rpm] | Bare Mass [gm] | Spun Mass [gm] | Liq. Mass [gm] | Thickness [um] | Unc.± [um] |
|-------------------|----------------|-------------------|-------------------|-------------------|-------------------|---------------|
| 1.1 | 1000 | 14.000 | 15.506 | 1.506 | 137.0 | 7.38 |
| 1.2 | 5000 | 14.000 | 14.753 | 0.753 | 68.5 | 3.69 |
| 1.3 | 6000 | 14.000 | 14.502 | 0.502 | 45.7 | 2.46 |
| 1.4 | 5000 | 14.000 | 14.376 | 0.376 | 34.2 | 1.84 |
| 1.5 | 4000 | 14.000 | 14.301 | 0.301 | 27.4 | 1.48 |
| 1.6 | 3000 | 14.000 | 14.251 | 0.251 | 22.8 | 1.23 |

ENGINEERING DATA SHEET

Excel Spreadsheet
Sheet: Spin Coating Data
File : ME 347 - Spin Coating Project.xlsm



Sheet No. X - 5 Date
Prepared by: HCM 08/01/16
Checked by: XXX XX/XX/XX
Approved by: XXX XX/XX/XX

HYPOTHETICAL thicknesses for future EXPERIMENTS

Liquid: Mineral Oil Sub. Dia.: 100 [mm]
Viscosity: 15 [mPa-s] Time: 50 [sec]
Density: 920 [kg/m^3]

| Data No. [-] | Speed [rpm] | Bare Mass [gm] | Spun Mass [gm] | Liq. Mass [gm] | Thickness [um] | Unc.± [um] |
|-------------------|----------------|-------------------|-------------------|-------------------|-------------------|---------------|
| 1.1 | 1000 | 14.000 | 14.045 | 0.045 | 6.3 | 0.35 |
| 1.2 | 5000 | 14.000 | 14.023 | 0.023 | 3.1 | 0.18 |
| 1.3 | 6000 | 14.000 | 14.015 | 0.015 | 2.1 | 0.13 |
| 1.4 | 5000 | 14.000 | 14.011 | 0.011 | 1.6 | 0.11 |
| 1.5 | 4000 | 14.000 | 14.009 | 0.009 | 1.3 | 0.10 |
| 1.6 | 3000 | 14.000 | 14.008 | 0.008 | 1.0 | 0.09 |

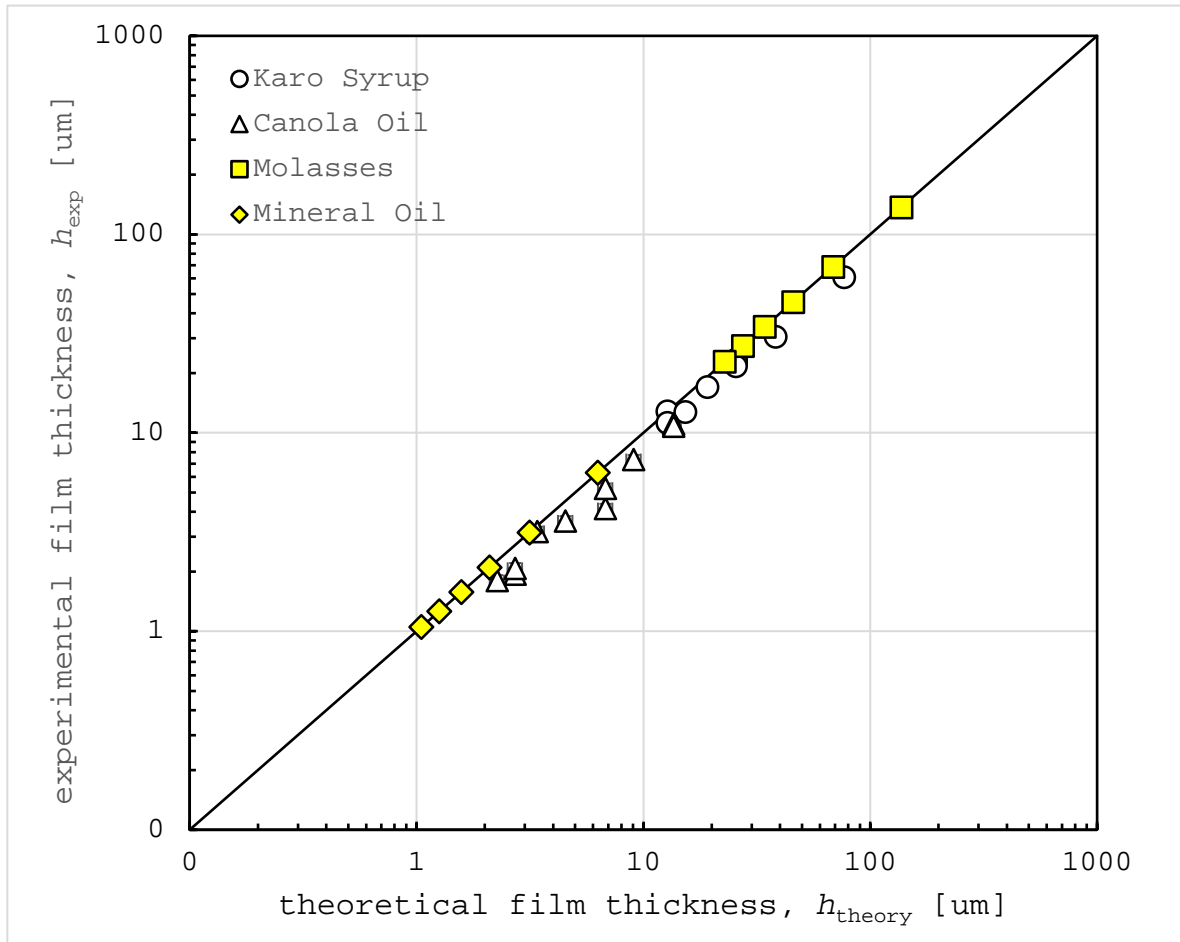
ENGINEERING DATA SHEET

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Sheet No. X - 8 Date
Prepared by: HCM 08/01/16
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Approved by: XXX XX/XX/XX

PLOT WITH EXPERIMENTAL DATA AND HYPOTHETICAL DATA



ENGINEERING DATA SHEET

Excel Spreadsheet
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File : ME 347 - Spin Coating Project.xlsm



Sheet No. X - 9 Date
Prepared by: HCM 08/01/16
Checked by: XXX XX/XX/XX
Approved by: XXX XX/XX/XX

HYPOTHETICAL thicknesses (THEORETICAL)

Liquid: Molasses Sub. Dia.: 100 [mm]
Viscosity: 10800 [mPa-s] Time: 50 [sec]
Density: 1400 [kg/m^3]

| Data No. [-] | Speed [rpm] | Thickness [um] | Unc.± [um] | % Diff [%] |
|-------------------|----------------|-------------------|---------------|---------------|
| 1.1 | 1000 | 137.0 | 9.97 | 0.000 |
| 1.2 | 2000 | 68.5 | 4.99 | 0.000 |
| 1.3 | 3000 | 45.7 | 3.32 | 0.000 |
| 1.4 | 4000 | 34.2 | 2.49 | 0.000 |
| 1.5 | 5000 | 27.4 | 1.99 | 0.000 |
| 1.6 | 6000 | 22.8 | 1.66 | 0.000 |

